CLAIMS

1. Manufacturing method for obtaining high-temperature components (10) for gas turbines, characterized in that it comprises at least incorporation of an element or insert (14) in a main body (12) of said component (10), said element or insert (14) having mechanical properties able to withstand stresses to which said component (10) is subject in a zone where the said element or insert (14) is arranged.

2. Manufacturing method according to Claim 1, characterized in that it envisages at least one fixing process for joining said element or insert (14) to said main body (12) of said component (10).

3. High-temperature component (10) for gas turbines, characterized in that it is obtained by incorporating at least one element or insert (14) in a main body (12) of said component (10), said element or insert (14) having mechanical properties able to withstand stresses to which said component (10) is subject in a zone where said element or insert (14) is arranged.

4. Component (10) according to Claim 3, characterized in that said element or insert (14) is of the modular type.

5. Component (10) according to Claim 3, characterized in that said element or insert (14) is arranged in an inlet zone of a blade.

6. Component (10) according to Claim 3, characterized in that said element or insert (14) is arranged in an outlet zone of a blade.

7. Component (10) according to Claim 3, characterized in that said element or insert (14) is made of a material which is more resistant to high temperatures than the material of said main body (12).